



NATIONAL AERONAUTICS AND SPACE ADMINISTRATION

MSC INTERNAL NOTE NO. 71-FM-155

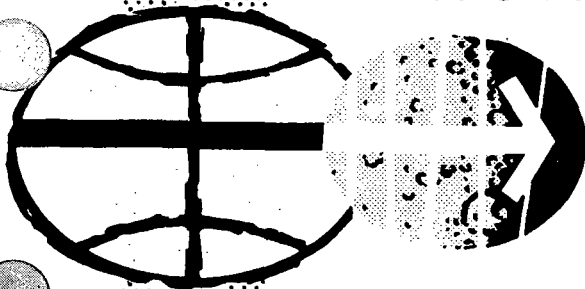
May 5, 1971

**CASE FILE
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**LUNAR SURFACE
NAVIGATION ANALYSIS**

Mathematical Physics Branch

MISSION PLANNING AND ANALYSIS DIVISION



**MANNED SPACECRAFT CENTER
HOUSTON, TEXAS**

Sgt- 66497

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PROJECT APOLLO

LUNAR SURFACE NAVIGATION ANALYSIS

By Robert T. Savely and J. Bruce Williamson
Mathematical Physics Branch

May 5, 1971

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LUNAR SURFACE NAVIGATION ANALYSIS

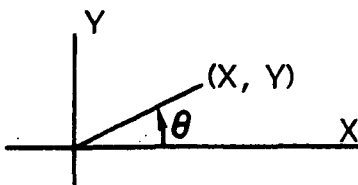
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INTRODUCTION

This study was performed in order to obtain an indication of how accurately the position of the crew could be obtained by using a line-of-sight device such as a sun compass or the ROVER television camera and the prominent mountains such as Hadley and Hadley δ . The targets selected for this study will be revised when the Mapping Sciences Laboratory has completed a study of the panoramic view. The study shows that with errors of 1° (noise only) the crew position can be obtained to $\sigma_\phi = 850$ feet and $\sigma_\lambda = 400$ feet using Hadley and Hadley δ . The addition of West Mountain reduced the errors to $\sigma_\phi = 600$ feet and $\sigma_\lambda = 400$ feet.

ANALYSIS AND RESULTS

The standard WLS method was used where

$$\theta = \tan^{-1} \frac{Y}{X}$$


$$\Delta\theta_i = \frac{\partial\theta_i}{\partial X} \Delta X + \frac{\partial\theta_i}{\partial Y} \Delta Y$$

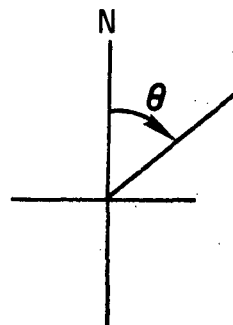
$$\frac{\partial\theta_i}{\partial X} = -Y/(X^2 + Y^2) = a_i, \quad \frac{\partial\theta_i}{\partial Y} = X/(X^2 + Y^2) = b_i$$

$$\begin{pmatrix} \sigma_X^2 & \sigma_{XY} \\ \sigma_{XY} & \sigma_Y^2 \end{pmatrix} = \alpha \begin{pmatrix} \Sigma a_i^2 & \Sigma a_i b_i \\ \Sigma a_i b_i & \Sigma b_i^2 \end{pmatrix}^{-1}$$

where $\alpha = \sigma_\theta^2 \text{ (rad)}^2$

The following X and Y coordinates were estimated from the map.

Mountain	X, km	Y, km	θ , deg
Hadley	14	17	45
Hadley δ	4	-15	165
West	-26	-6	260
300°	-18	12	300
100°	25	-5	100



The following results were obtained using the preceding data and the program in the appendix.

Hadley and Hadley δ

No. of marks per target	σ_{ϕ} , ft	σ_{λ} , ft
1	1500	700
2	1050	500
3	850	400
4	750	350

Hadley, Hadley δ , and West

No. of marks per target	σ_{ϕ} , ft	σ_{λ} , ft
1	1000	700
2	700	500
3	600	400

Hadley, Hadley δ , West, and 300°

No. of marks per target	σ_{ϕ} , ft	σ_{λ} , ft
1	1000	850
2	750	650
3	450	400

All 5 targets

No. of marks per target	σ_{ϕ} , ft	σ_{λ} , ft
1	800	800
2	500	450
3	400	400

Hadley, Hadley δ , and 300°

No. of marks per target	σ_{ϕ} , ft	σ_{λ} , ft
1	1000	700
2	700	500
3	600	400

CONCLUSIONS

The requirement of near targets with the optimal separation of 90° is intuitively observed in the results as well as the value of taking three marks on each target to smooth the data and to provide a verification mark in the event that one of the three marks is inaccurate. Note that since the analysis is linear when the error models for the devices are determined the results can be scaled. It should be kept in mind that the results are noise only and that the study will be repeated with angle bias and target location error. The results of this study show that with a 1° sighting error (noise only) that the greatest accuracy that can be expected is 400 feet in latitude and longitude.

and the results are

1954-1955

APPENDIX I- LUNAR SURFACE NAVIGATION PROGRAM

LANGUAGE: COMPUTER COMPLEX XTRAN

1CONTINUE

P1=0.

P2=0.

P3=0.

DISPLAY NAVIGATION NOISE ONLY ERROR ANALYSIS\$
 DISPLAY SINPUT FLAG: 0 FOR NO APRIORI, 1 FOR APRIORI\$

ACCEPT IFLAG1

IF(IFLAG1),2,

WRITE (1,100)

100 FORMAT(SAPRIORI MATRIX HAS FORMS\$,/, \$ / A B /\$,/,
 \$ / B C /\$,/, SINPUT A, B, C(RETURN)\$)

ACCEPT A1,A2,A3

DET=A1*A3-A2*A2

P1=A3/DET

P2=-A2/DET

P3=A1/DET

2CONTINUE

DISPLAY SINPUT TARGET LOCATION AS DISPLACEMENT EAST (X), DISPLACEMENT
 NORTH (Y)\$

READ(0,102) X,Y,IFLAG2

R=X*X+Y*Y

DTHDX=-Y/R

DTHDY=X/R

P1=P1+DTHDX*DTHDX

P2=P2+DTHDX*DTHDY

P3=P3+DTHDY*DTHDY

IF(IFLAG2),2,

DET=P1*P3-P2*P2

C1=P3/DET

C2=-P2/DET

C3=P1/DET

SIGX=SQRT(C1)

SIGY=SQRT(C3)

RHO=C2/(SIGX*SIGY)

WRITE(1,101) C1,C2,RHO,C3,SIGX,SIGY

ACCEPT IFLAG3

IF(IFLAG3),2,1

STOP

101FORMAT(2E15.8/2E15.8/\$SIGMA X =\$,E15.8,\$SIGMA Y =\$,E15.8//
 SINPUT FLAG: -1 TO END RUN, 0 TO PROCEED, +1 TO REINITIALIZES)

102FORMAT(2E15.8,13)

END